

Structures of Shear Turbulence

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Historical Overview

Adapted from

"A Brief History of Boundary Layer Structure Research," S.J. Kline 1997

- Mean Flow Era 1883-1936
(Reynolds, Prandtl, von Karman,...)

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- **CFD** Era 1986-

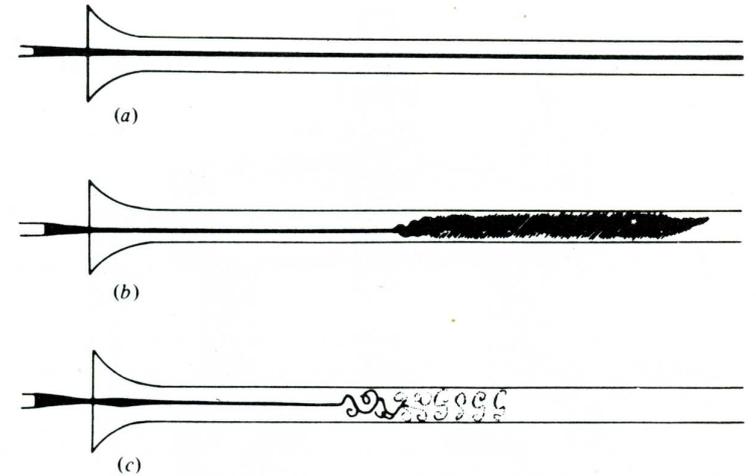
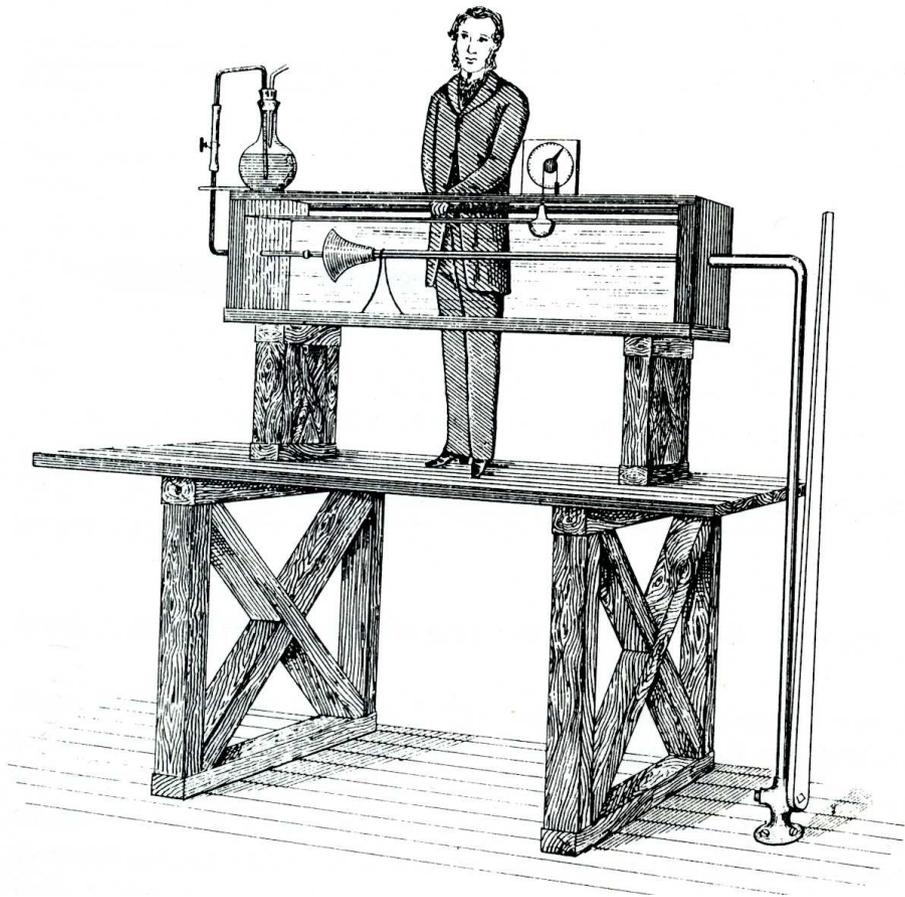
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- Dynamical Era: Self-Sustaining Process
Exact Coherent Structures
Periodic Solutions,...

Reynolds' Pipe Flow Experiments



Osborne Reynolds, Manchester 1883

Reynolds' Key Observations

- *"... at some point in the tube, always at a considerable distance from the intake, the colour band would all at once mix up with the surrounding water..."*
- **Linear Stability - Nonlinear instability:**
"... the critical velocity was very sensitive to disturbance in the water before entering the tubes.... This at once suggested the idea that the condition might be one of instability for disturbances of a certain magnitude and stability for smaller disturbances".

Mean Flow Era Concepts

- Reynolds similarity: $R = \frac{UL}{\nu}$
- Mean flow + fluctuations (Reynolds 1894)

$$\mathbf{v}(x, y, z, t) = \bar{U}(y)\hat{\mathbf{x}} + \mathbf{u}(x, y, z, t)$$

- Reynolds Stress: $\tau = \nu \frac{d\bar{U}}{dy} - \overline{uv}$
- Law of the wall (Prandtl 1925) (wall unit scaling)
- log law, velocity defect law (von Karman 1930)

Mean Flow Era Concepts (2)

- 'eddy-viscosity': $-\overline{uv} \approx \tilde{\nu} \frac{d\overline{U}}{dy}$
- 'Mixing length' l : $\tilde{\nu} \approx u_* l, \quad u_* = |\overline{uv}|^{1/2}$
$$\Rightarrow \tilde{\nu} = l^2 \left| \frac{d\overline{U}}{dy} \right|$$
- Turbulence= random interactions of "eddies"
- Richardson Cascade (1922)
*"Big whorls have little whorls, Which feed on their velocity,
Little whorls have lesser whorls, And so on to viscosity"*

momentum transport \longrightarrow energy cascade

Statistical Era

- Turbulent motion 'not as random' as molecular motion
 $\langle u(x)v(x) \rangle \rightarrow$ 2-point correlation $\langle u_i(x+r)u_j(x) \rangle$

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GI Taylor 1935, **grid turbulence**; Karman-Howarth 1938

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- Kolmogorov: energy dissipation rate per unit mass \mathcal{E}

$$\langle (u(x+r) - u(x))^n \rangle \propto (\mathcal{E}r)^{n/3}$$

$$\text{if } \eta = (\nu^3/\mathcal{E})^{1/4} \ll r \ll L \quad (\text{inertial range})$$

(OK for $n = 2$, departures for $n > 2$, intermittency...)

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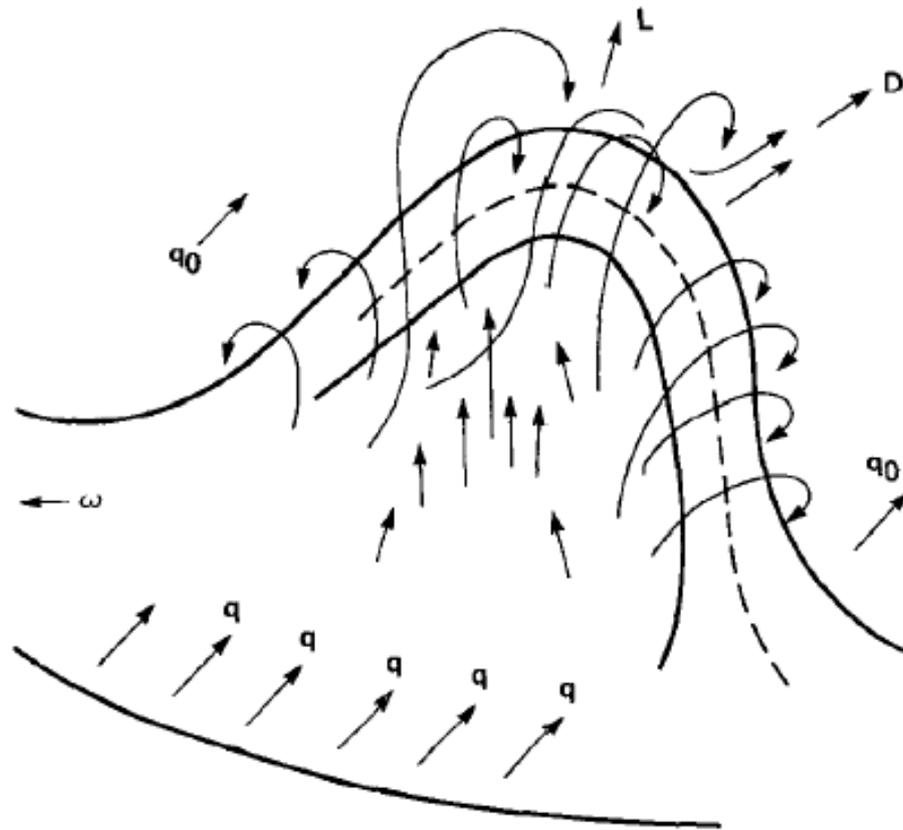
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- **No momentum transport, only cascade of energy!**

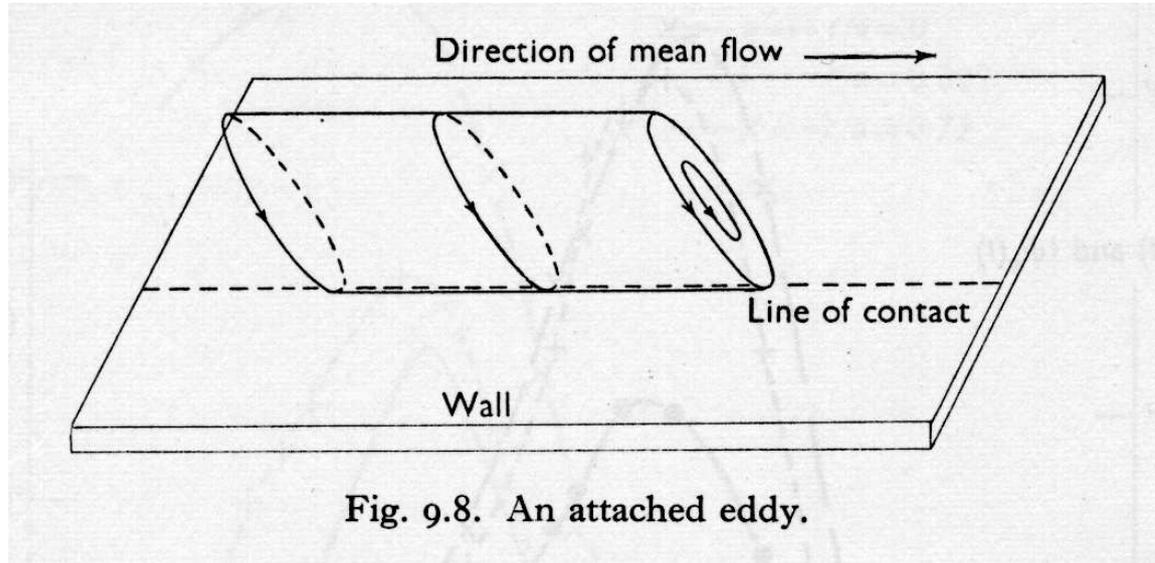
‘Horseshoe’ structure as self-consistent ‘turbulence molecule’
“optimized” for vortex stretching by mean shear



Theodorsen, 1952

Structural Era

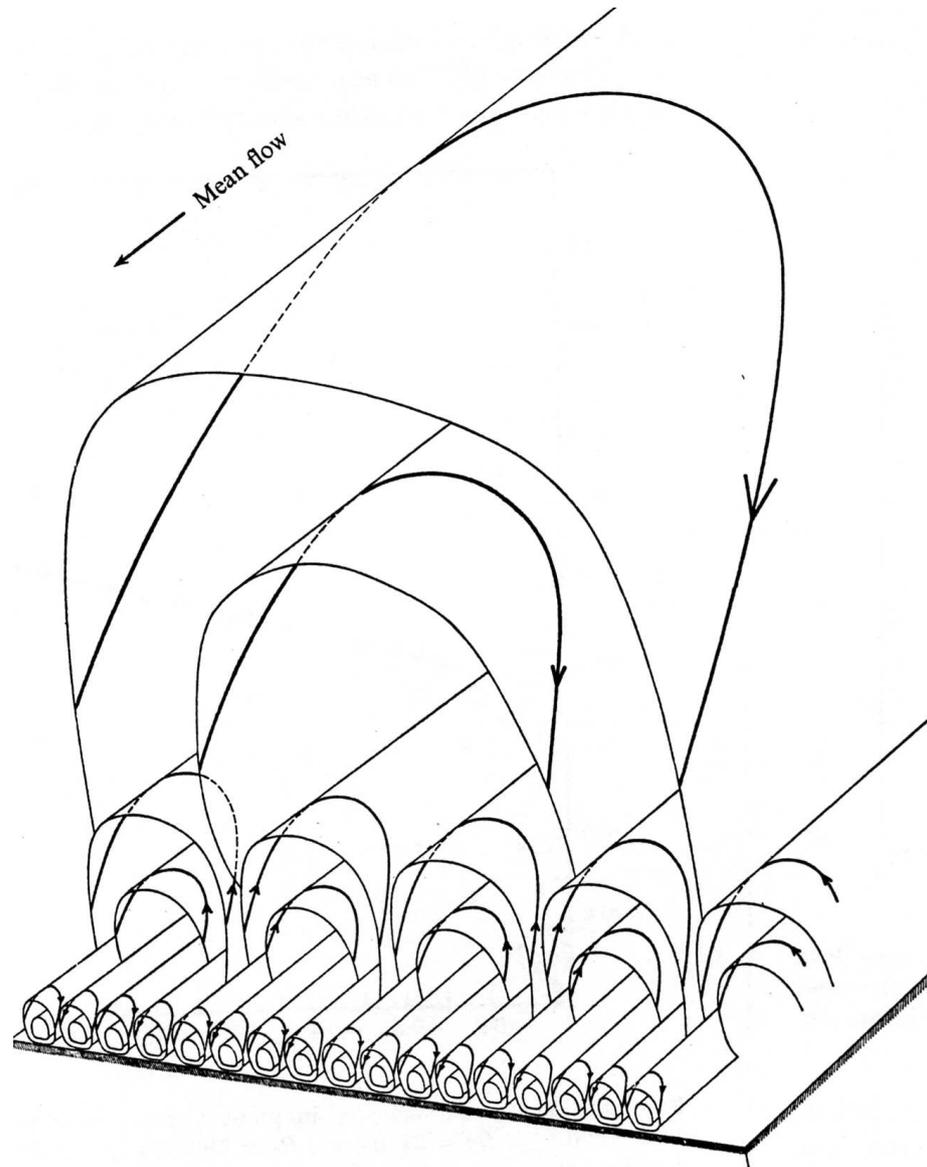
2-point correlations \longrightarrow Structures



Townsend 1956 'The Structure of Turbulent Shear Flows'

- *'Outline of a theory of turbulent convection'* Malkus 1954
'Outline of a theory of turbulent shear flows' Malkus 1956
- Marginal Stability, Optimum transport
- Upper bound theory
(Howard, Busse, Malkus & Lerley & L.M. Smith,...)
- Background field approach (Doering & Constantin, Kerswell...)

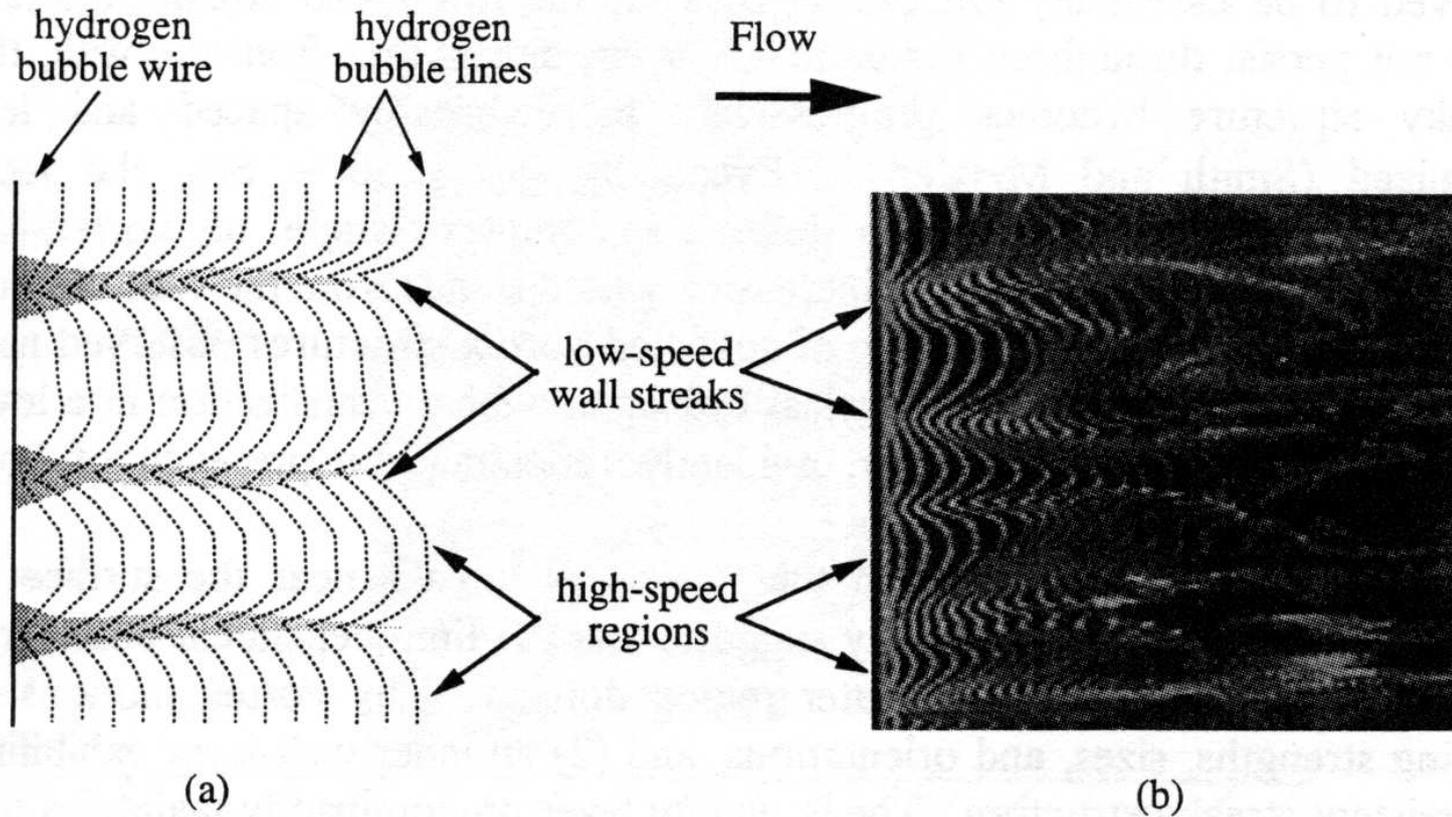
'Optimum' transport structure



Busse, JFM 1970

Visualization Era: Streaks

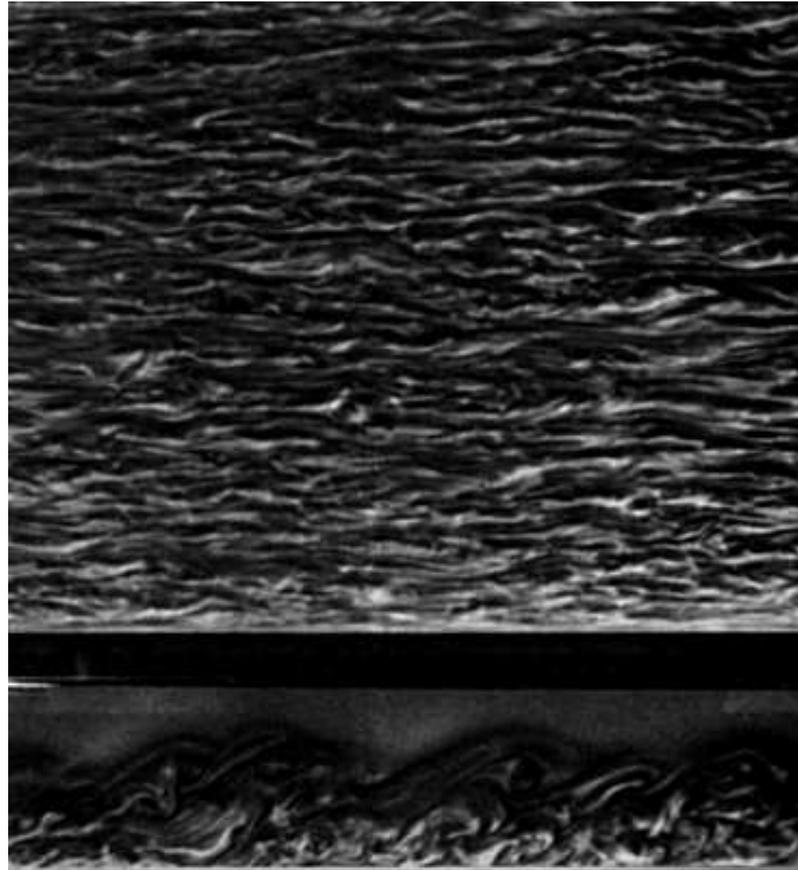
Streaks with 100^+ spacing



*Kline, Reynolds, Schraub & Runstadler, JFM 1967
(diagram from Smith & Walker, 1997)*

Visualization Era: Streaks everywhere!

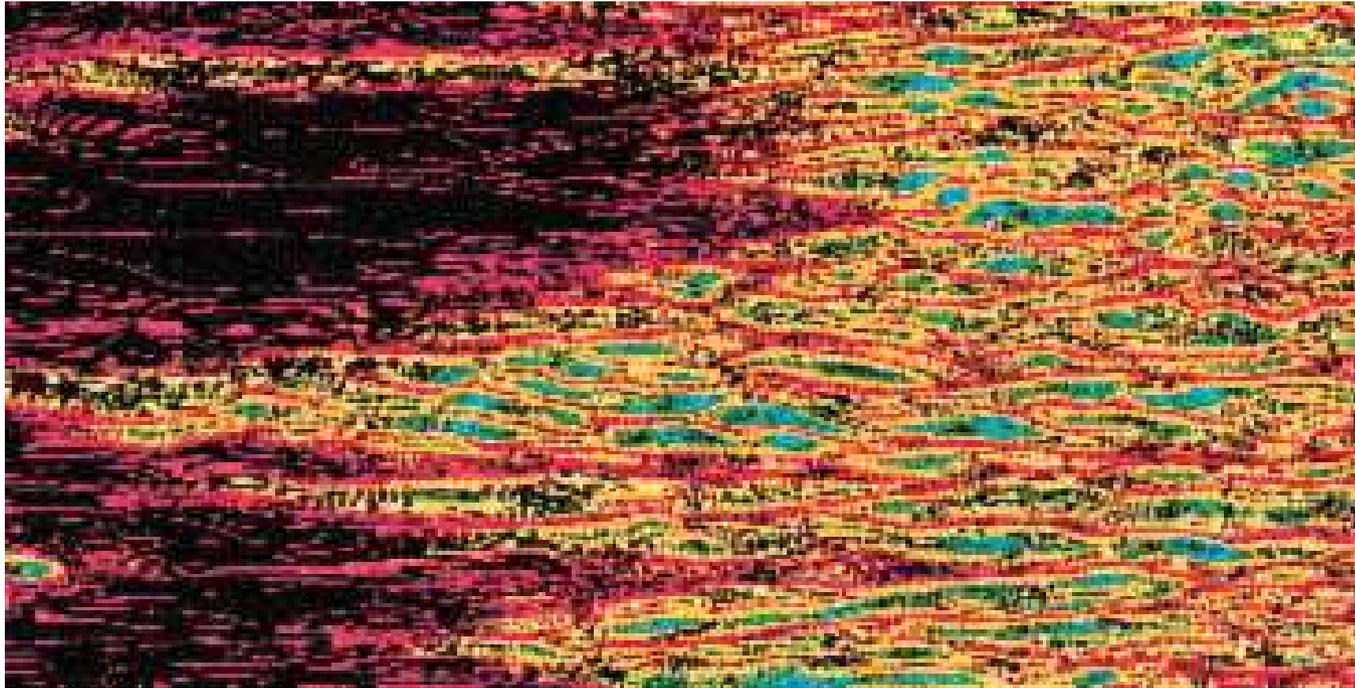
Streaks in Turbulent Boundary Layer



Cantwell, Coles & Dimotakis, JFM 1978

Boundary Layer Transition: Streaks!

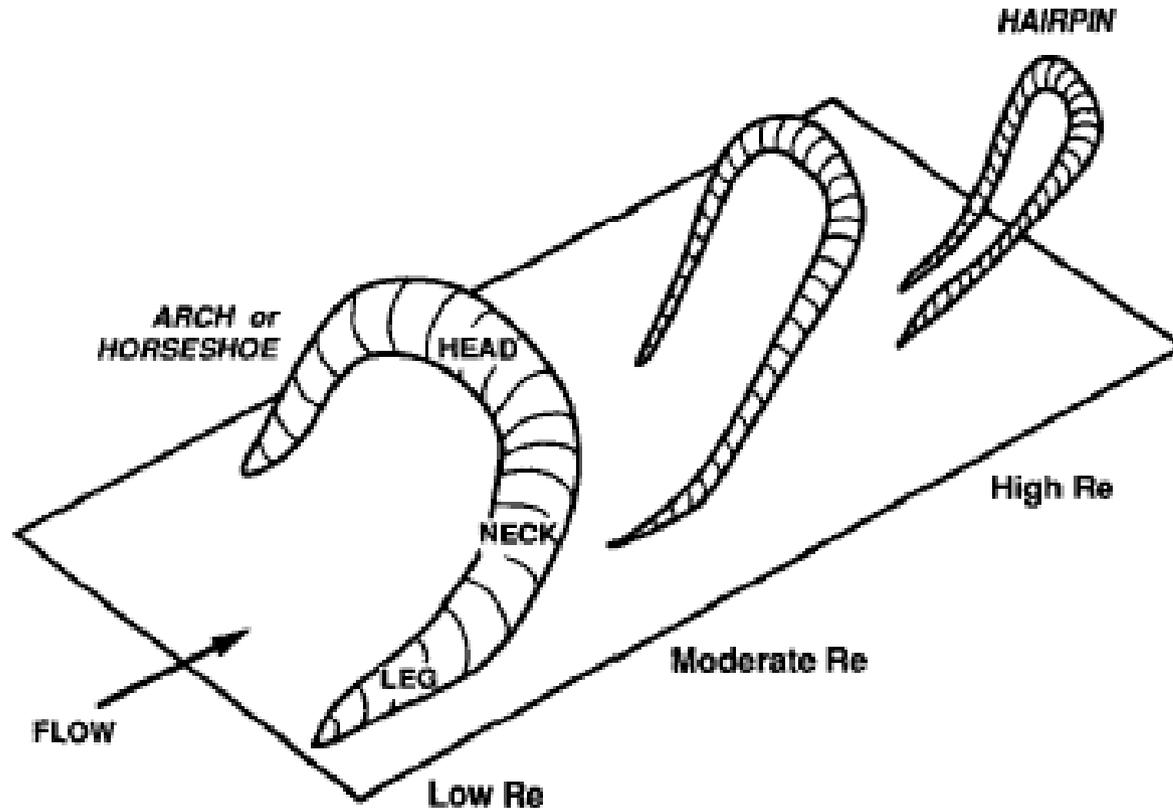
DNS of 'Natural' transition



Rai & Moin, AIAA 1991

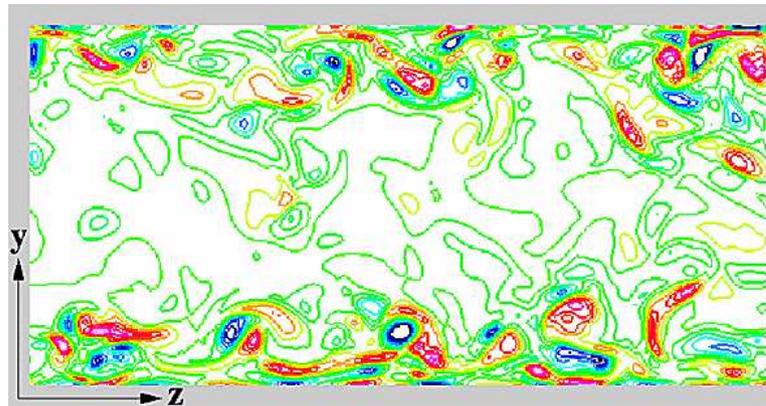
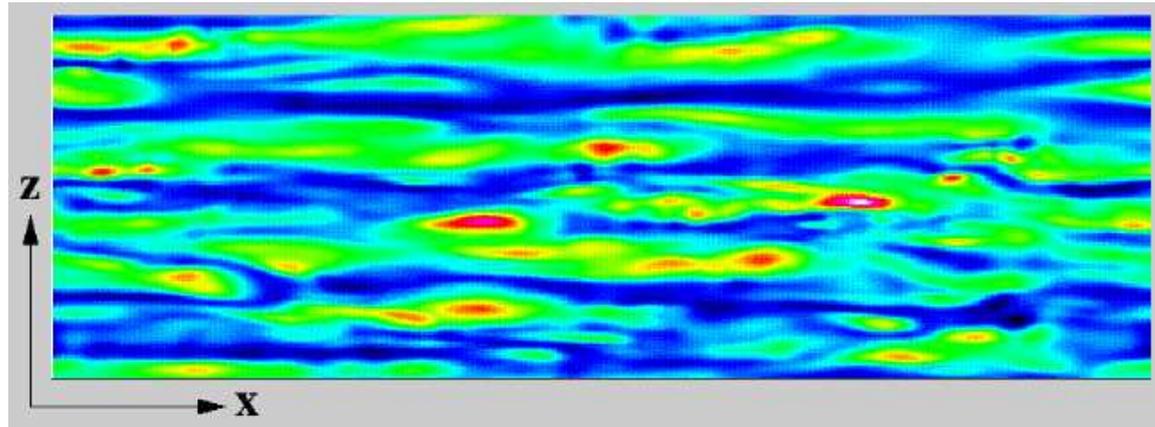
Visualization Era

Typical **vortex** structure: Theodorsen's horseshoe!



Head & Bandyopadhyay, JFM 1981

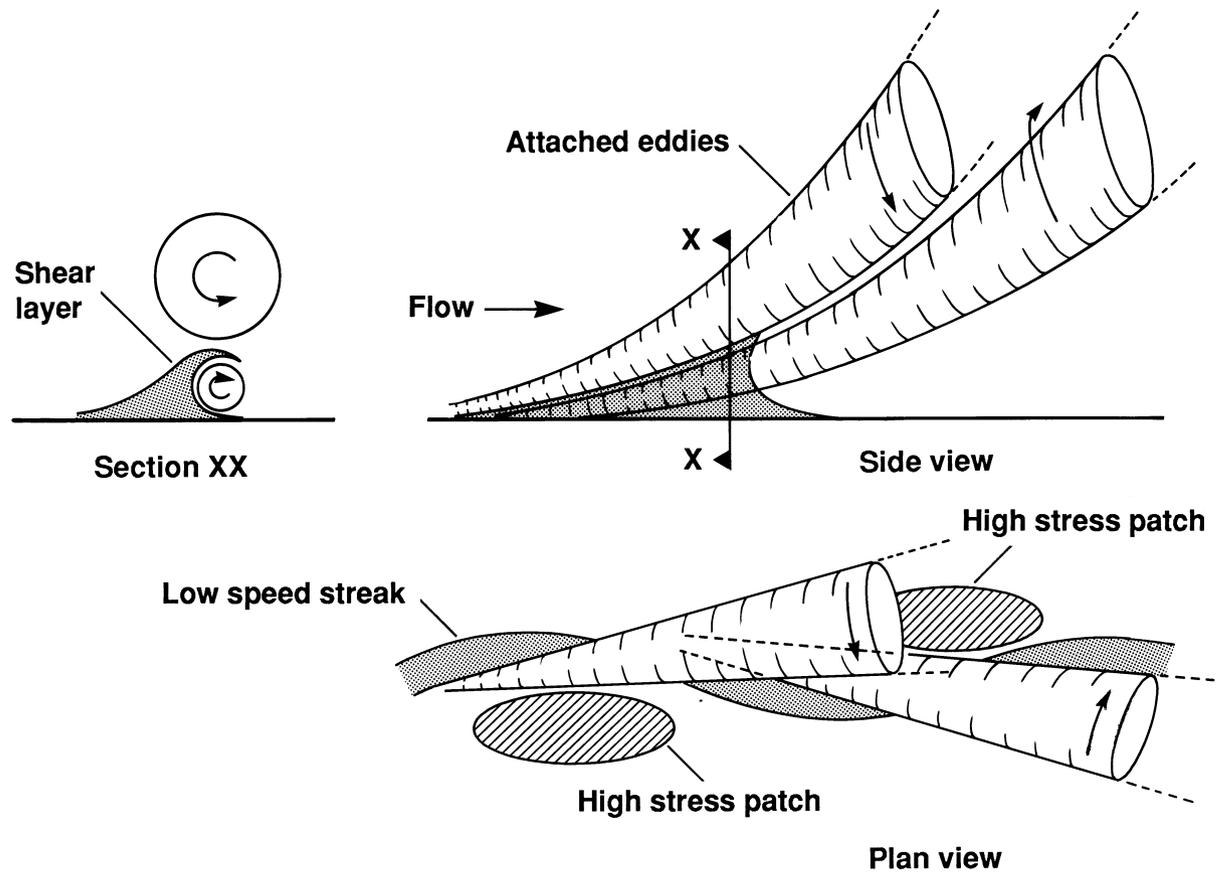
DNS of Turbulent Channel Flow



Kim, Moin & Moser, JFM 1987

$$R_\tau \approx 180, R_m \approx 5600$$

DNS near-wall Structures



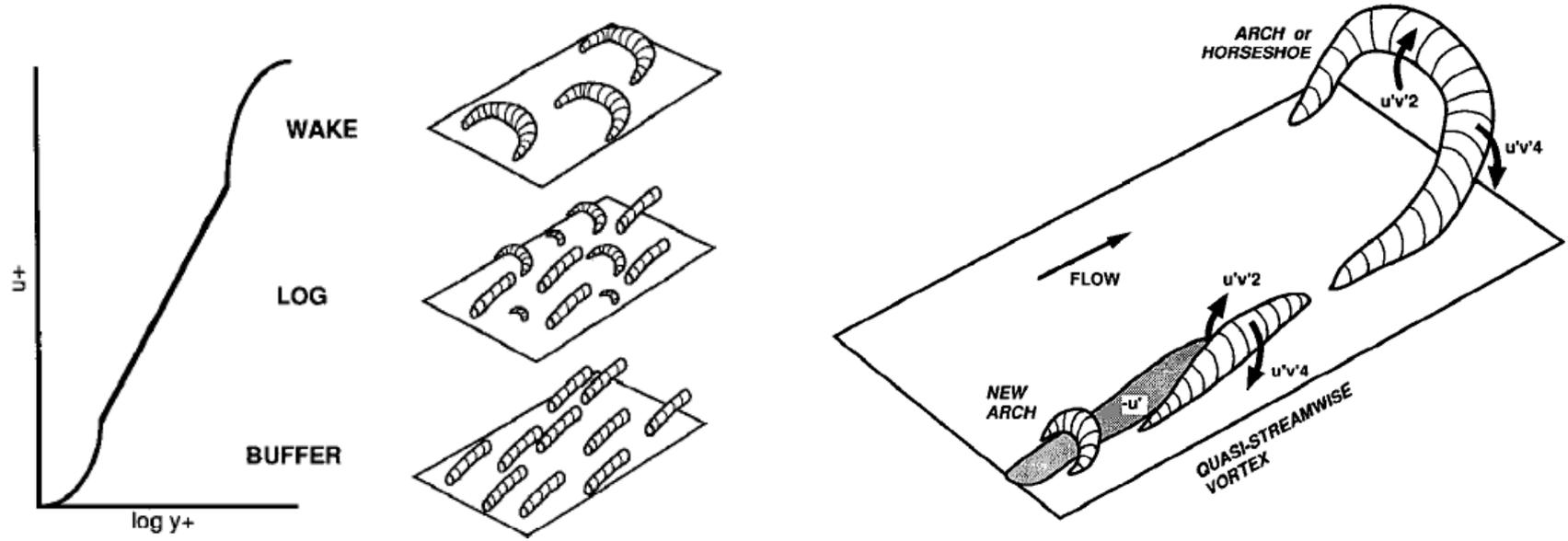
STRETCH

Asymmetric CS educed from KMM channel data at $R_\tau = 180$

Derek Stretch, 1990

Visualization/DNS Era

baguettes and croissants! not spaghetti!



3D 'Inverse' cascade: buffer layer → outer flow

Steve Robinson, ARFM 1991

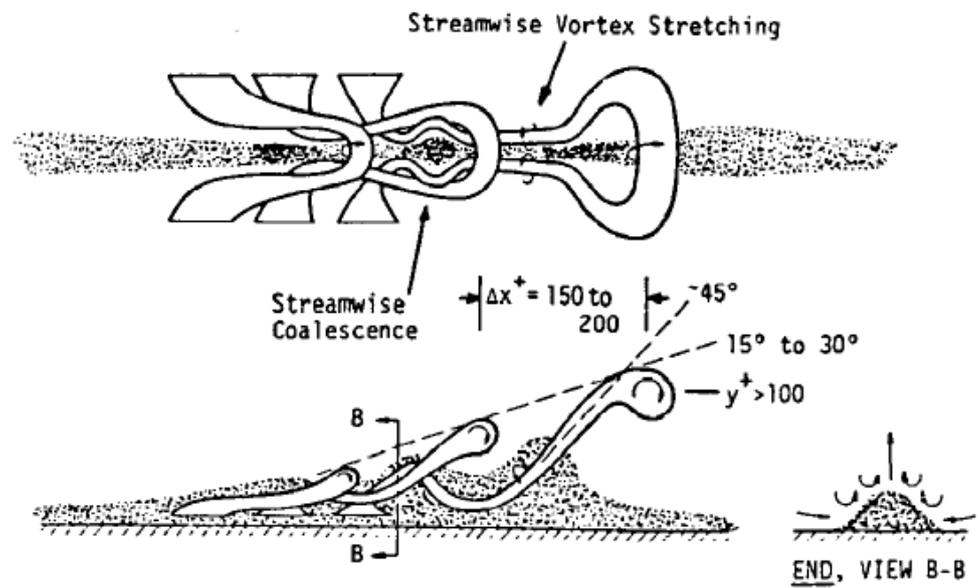
...buffer layer to outer space!



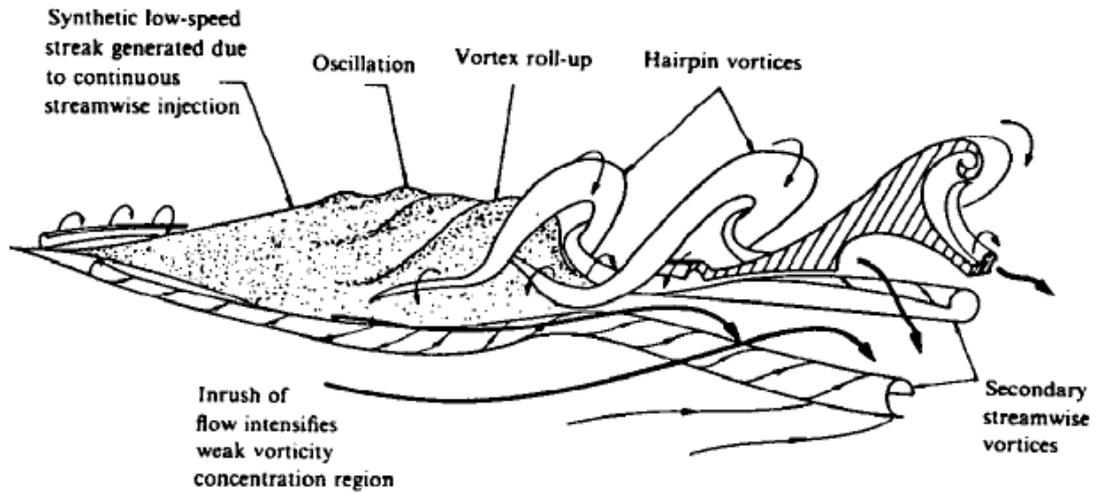
STS 114: Return to Flight, launched July 26, 2005...

Mechanistic Era

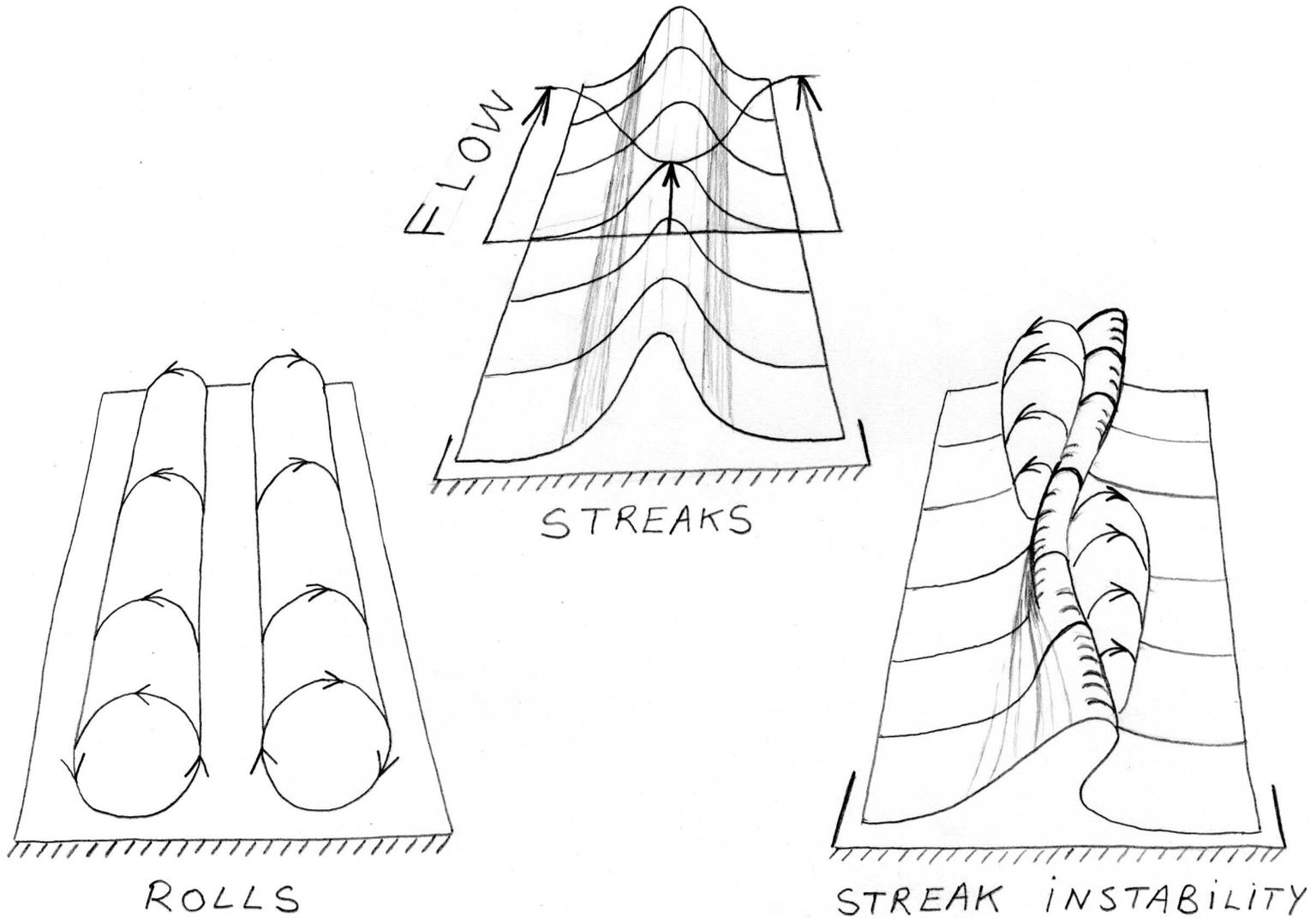
Acarlar & Smith
JFM 1987
 (synthetic streak
 regeneration)
 + Benney 1984



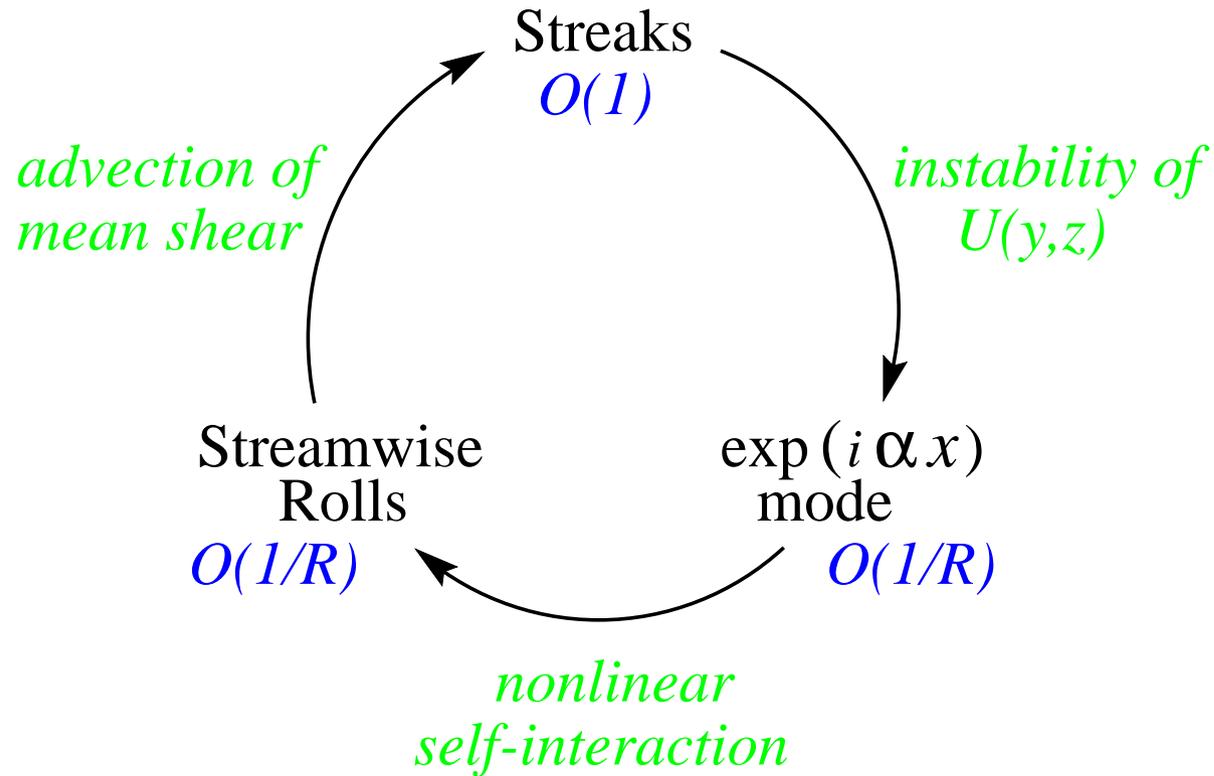
d) vortex ejection, stretching and interaction



Cartoon of Self-Sustaining Process



Self-Sustaining Process



Waleffe, Stud. Applied Math 1995, Phys. Fluids 1995, 1997

SSP as computational method, PRL 1998, JFM 2001, PoF 2003

SSP as Computational Method

- Add artificial roll forcing $O(1/R^2)$ to Navier-Stokes Equations

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- 1D-1C Laminar flow \longrightarrow 2D-3C "streaky flow"

$$\mathbf{v} = U(y, z)\hat{\mathbf{x}} + \frac{1}{R} [V(y, z)\hat{\mathbf{y}} + W(y, z)\hat{\mathbf{z}}]$$

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- Track 3D-3C solution that bifurcates from marginally stable streaky flow

(Newton's method, continuation, huge nonlinear eigenvalue problem)

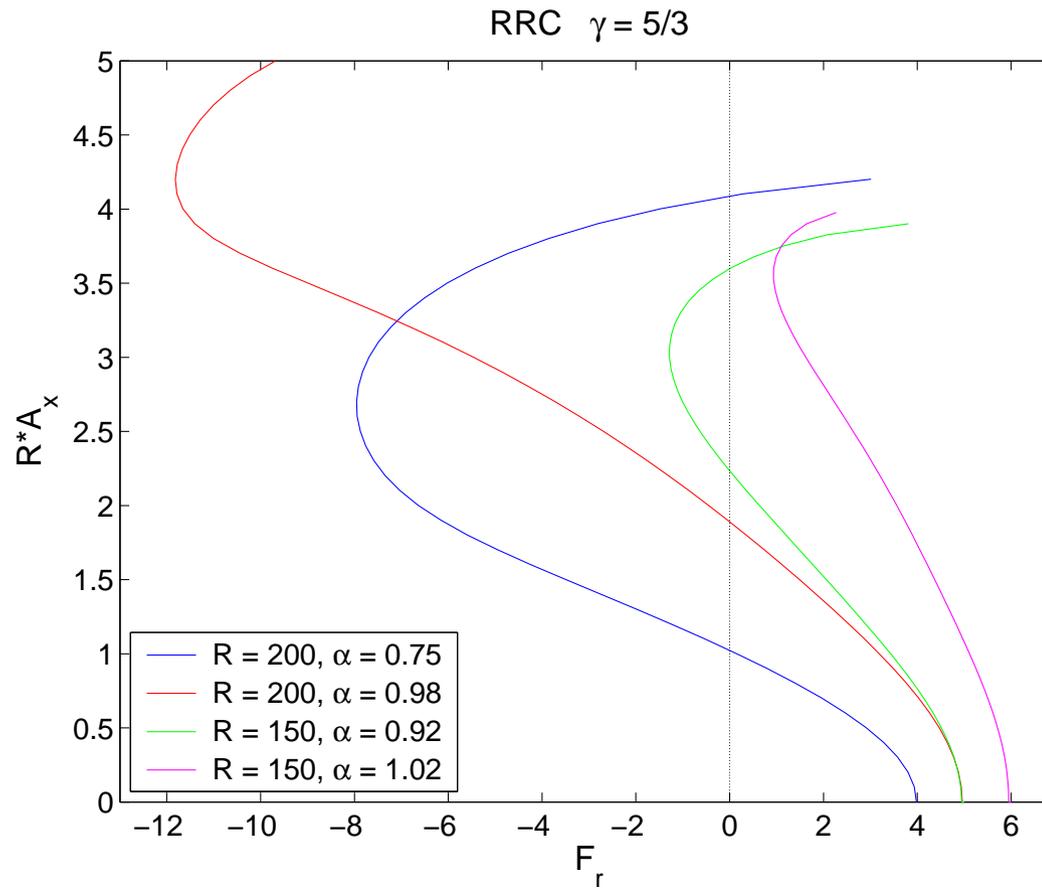
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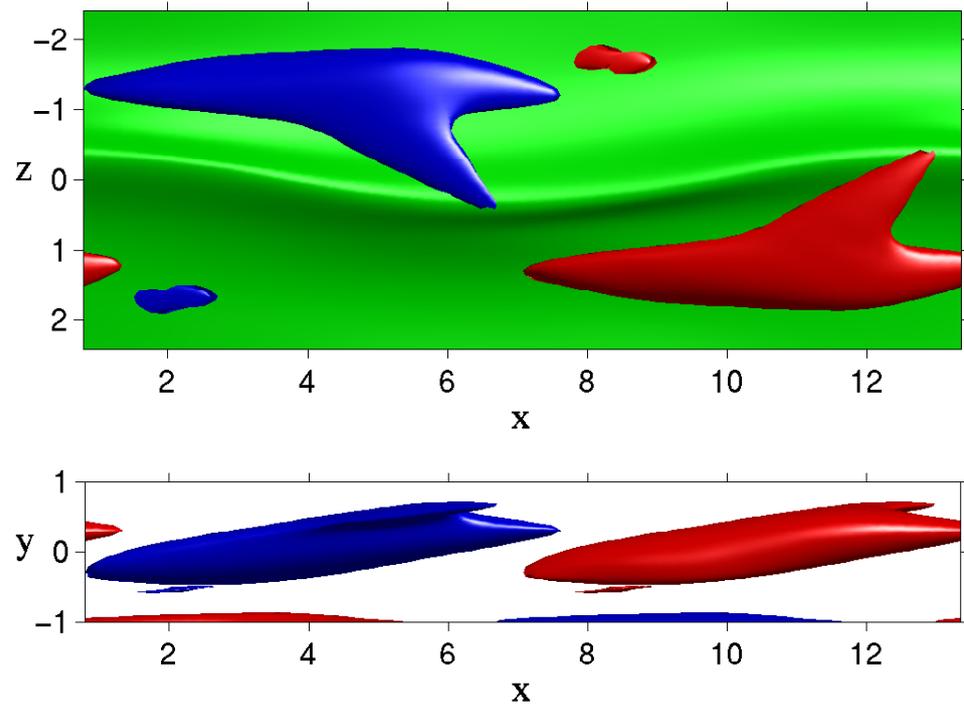
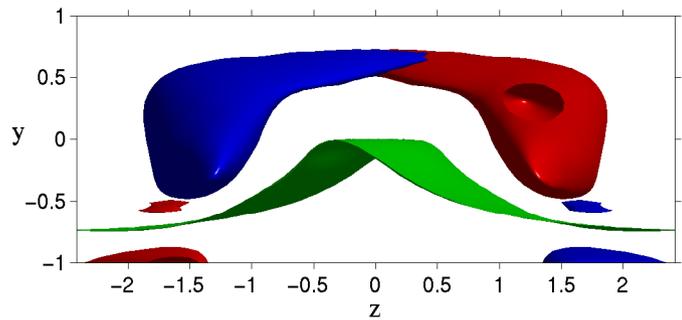
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- Continue till no need for artificial roll forcing

Bifurcation from streaky flow (PCF)



$$V_{\max} = \frac{F_r}{R}$$

Traveling Wave Solutions (1/2 PPF)



Waleffe, JFM 2001, PoF 2003

Traveling Wave Solutions

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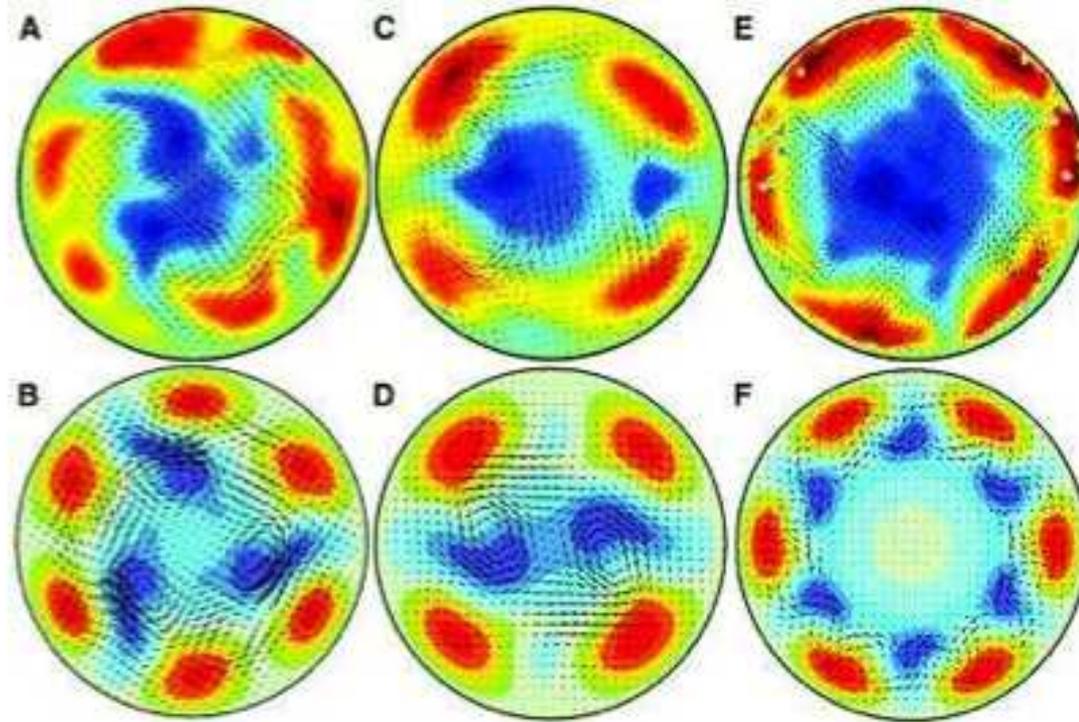
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- unstable...
- ... *a feature not a bug!* (more tomorrow!)
- generic:
 - Plane Couette flow (wall driven)
 - Plane Poiseuille flow (pressure driven)
 - free-slip, no-slip, any-slip
 - Pipe flow

Traveling Waves in Pipe Flow

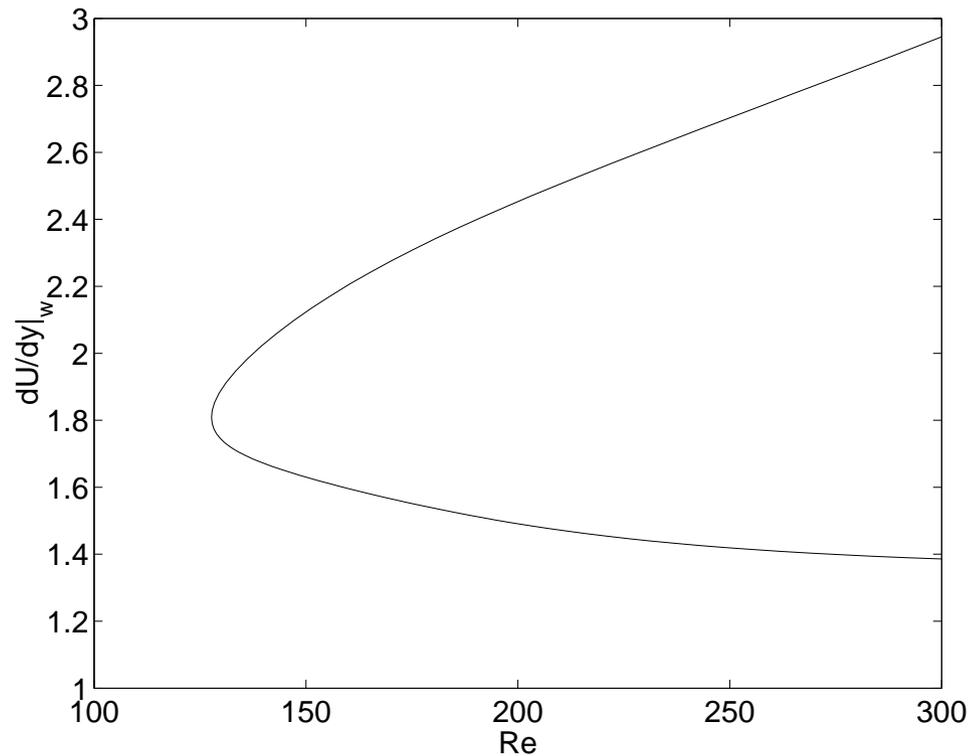


Faisst & Eckhardt, PRL 2003, Wedin & Kerswell JFM 2004

Hof *et al.* Science, Sept. 2004

ECS come in pairs

Bifurcation Diagram (saddle-node $F_r \equiv 0$):
upper and lower branches



Normalized wall shear rate (drag) in Plane Couette Flow vs R

- **upper** branch ECS \approx backbone of turbulence

capture 'statistics' pretty well (mean and rms profiles)

Waleffe PoF 2003, Jimenez *et al.* PoF 2005

- **lower** branch ECS \approx backbone of separatrix

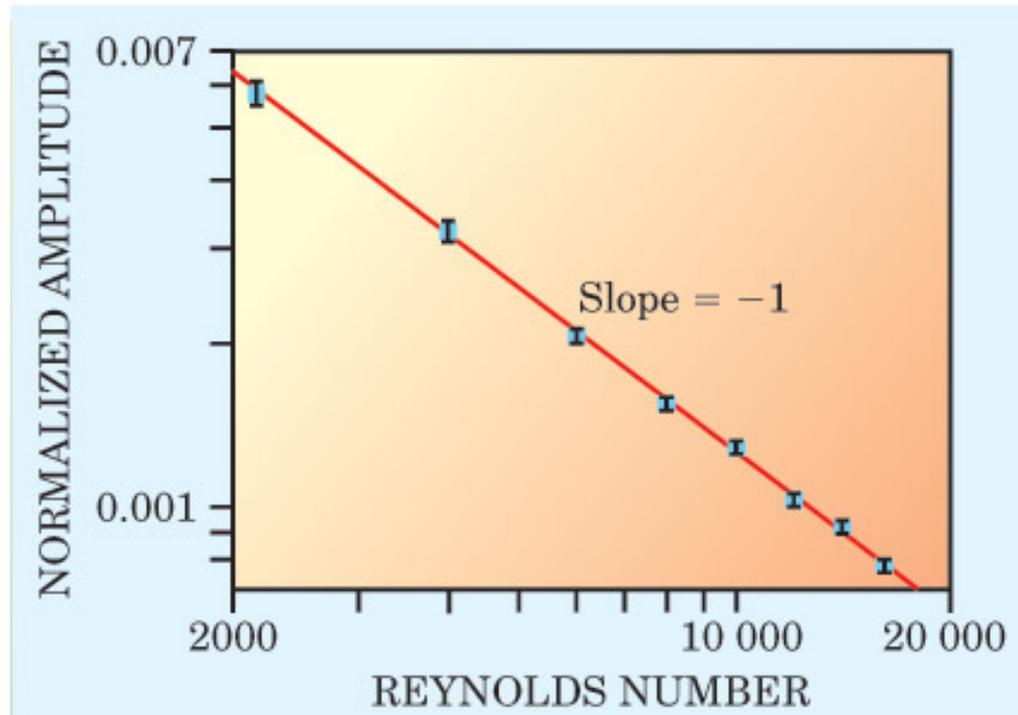
(laminar and turbulent separated by stable manifold(s) of lower branch(es))

- scaling of **lower** branch ECS \leftrightarrow transition threshold

Outstanding evidence for $1/R$ (Pipe flow)

Search and Discovery

Figure 2



Turbulence threshold scaling. The amplitude required to establish turbulence scales inversely with the Reynolds number of flow when the injected flux is normalized by the flux flowing through the pipe. (Adapted from ref. 2)

Hof, Juel & Mullin, PRL, Dec 2003; Physics Today Feb 2004

Fourier in $\theta = x - ct$ (traveling wave)

$$u(\theta, y, z) = u_0(y, z) + u_1(y, z)e^{i\theta} + u_2(y, z)e^{i2\theta} + \dots$$

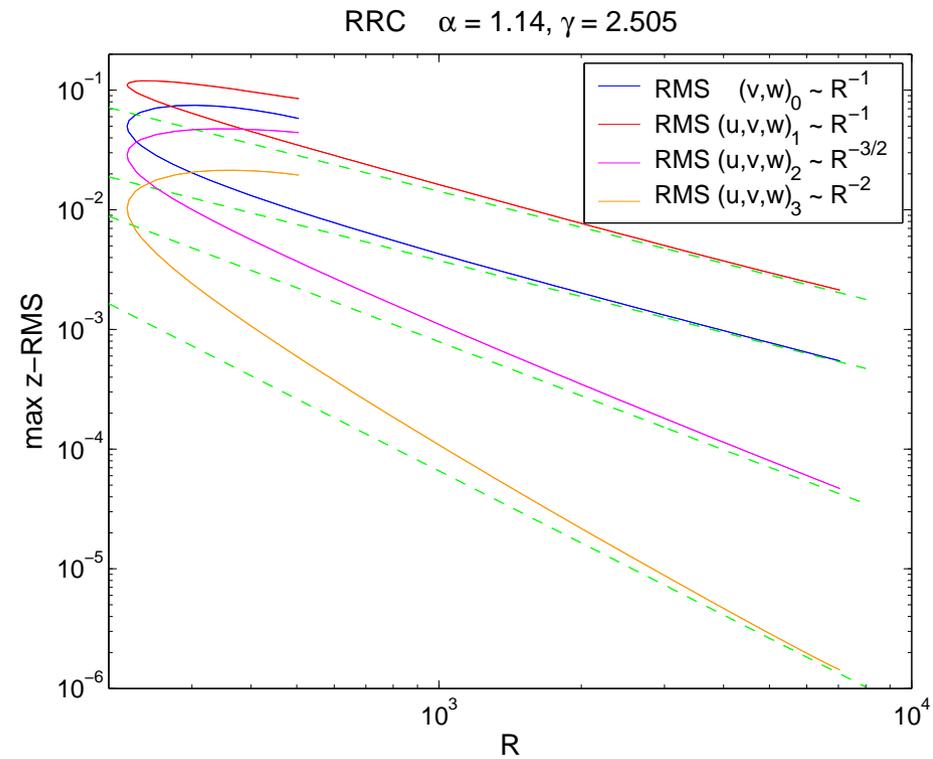
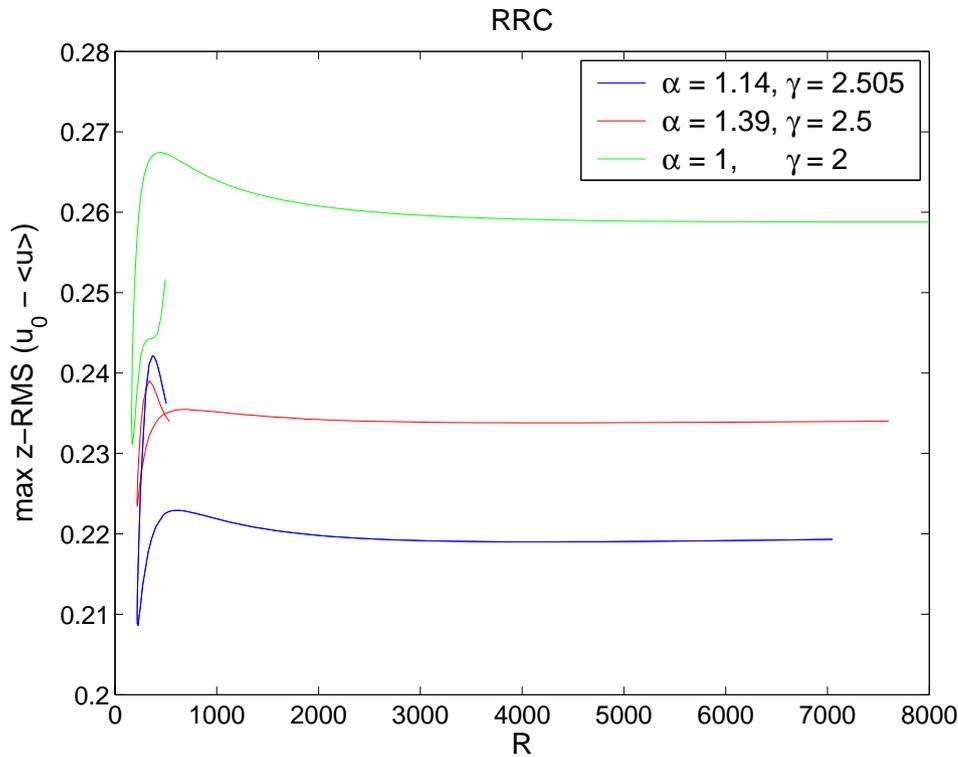
$$v(\theta, y, z) = v_0(y, z) + v_1(y, z)e^{i\theta} + \dots$$

$$w(\theta, y, z) = w_0(y, z) + w_1(y, z)e^{i\theta} + \dots$$

In SSP theory:

$$u_0(y, z) = O(1), \quad v_0, w_0 = O\left(\frac{1}{R}\right), \quad u_1, v_1, w_1 = O\left(\frac{1}{R}\right)$$

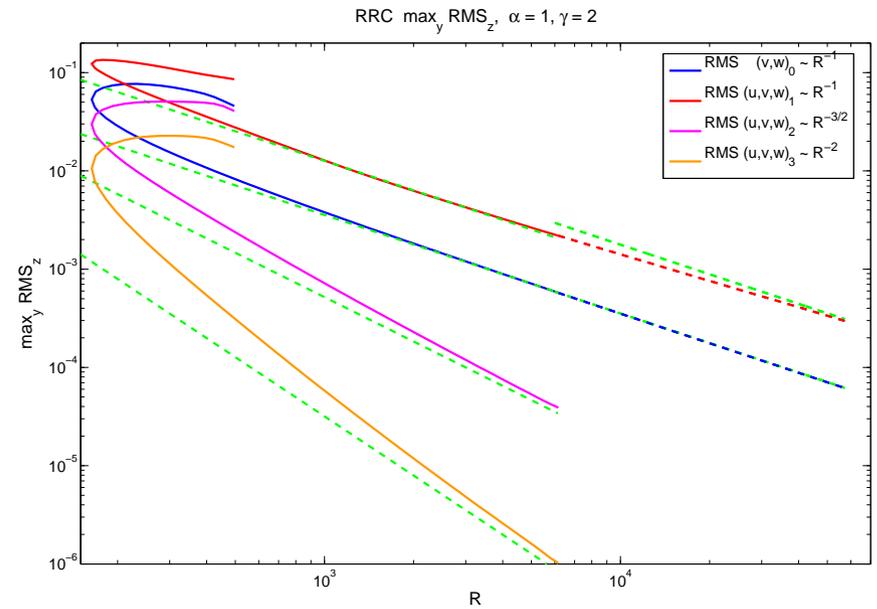
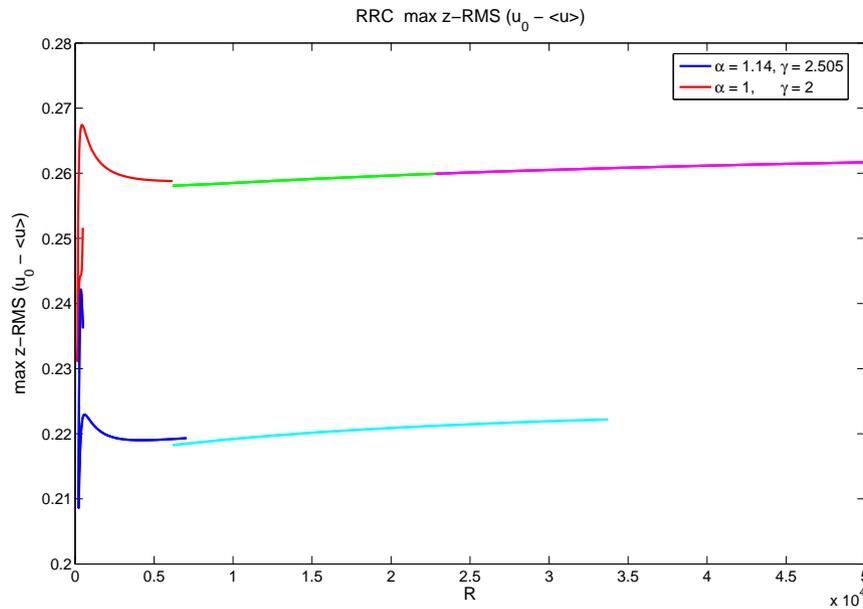
R -Scaling of harmonics (Rigid Rigid Couette)



$$\max_y RMS_z \left(u_0(y, z) - \bar{U}(y) \right), \quad (v_0, w_0), (u_1, v_1, w_1), \dots$$

Waleffe & Wang, 2004

R-Scaling dropping higher harmonics

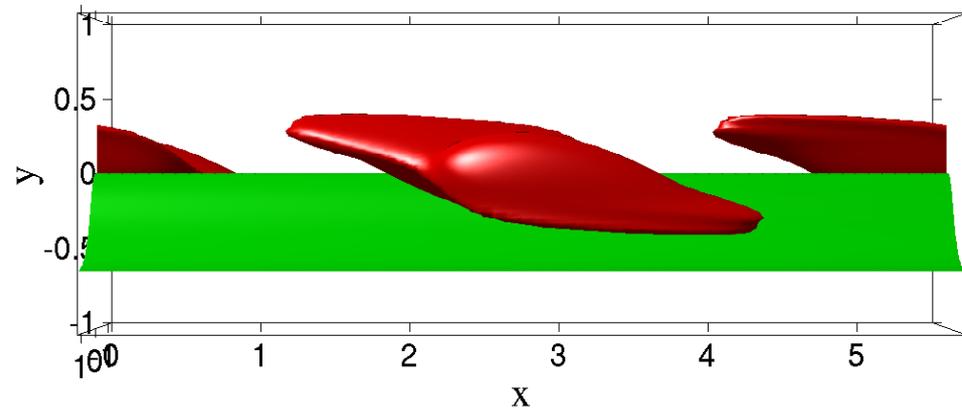
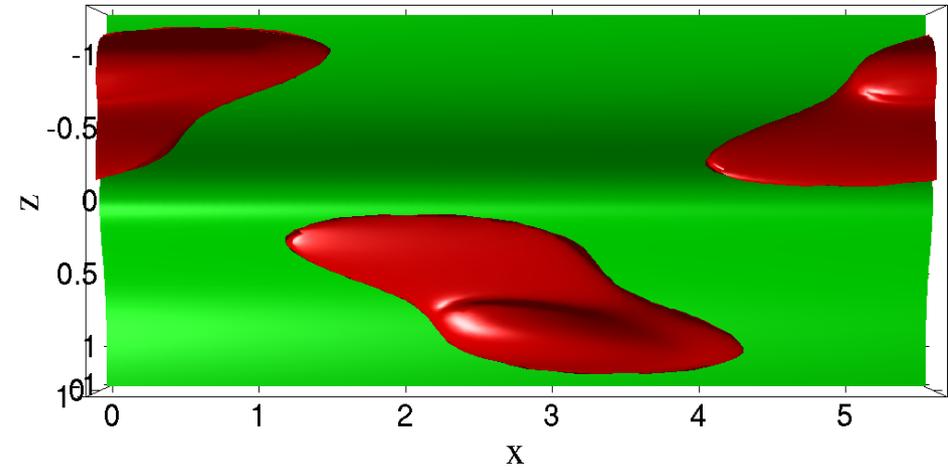
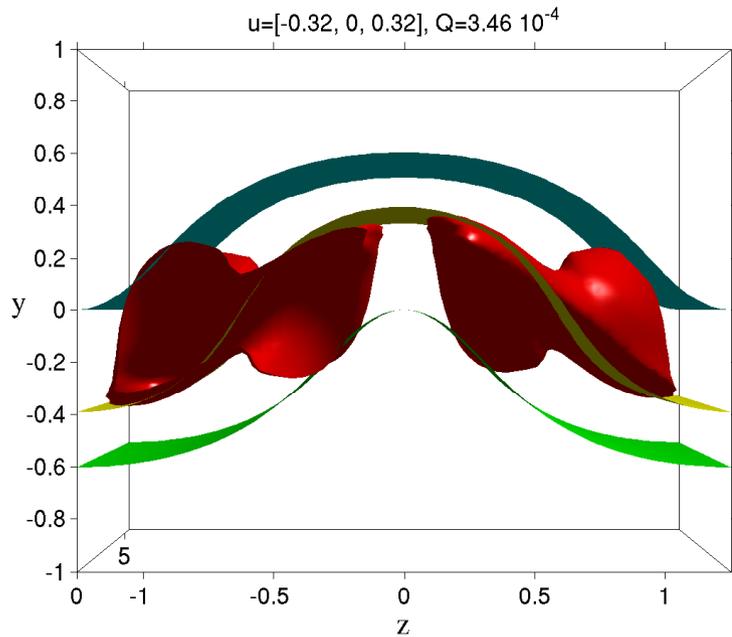


$$\max_y \text{RMS}_z \left(u_0(y, z) - \bar{U}(y) \right), \quad (v_0, w_0), (u_1, v_1, w_1), \dots$$

Waleffe & Wang, 2005

Trouble in the 1st harmonic?...

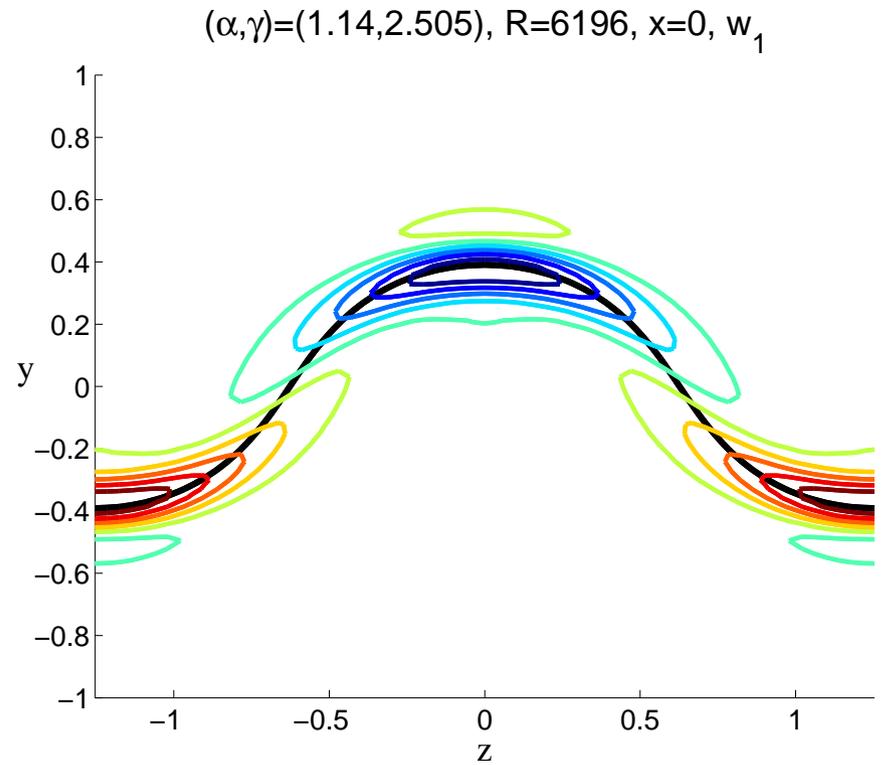
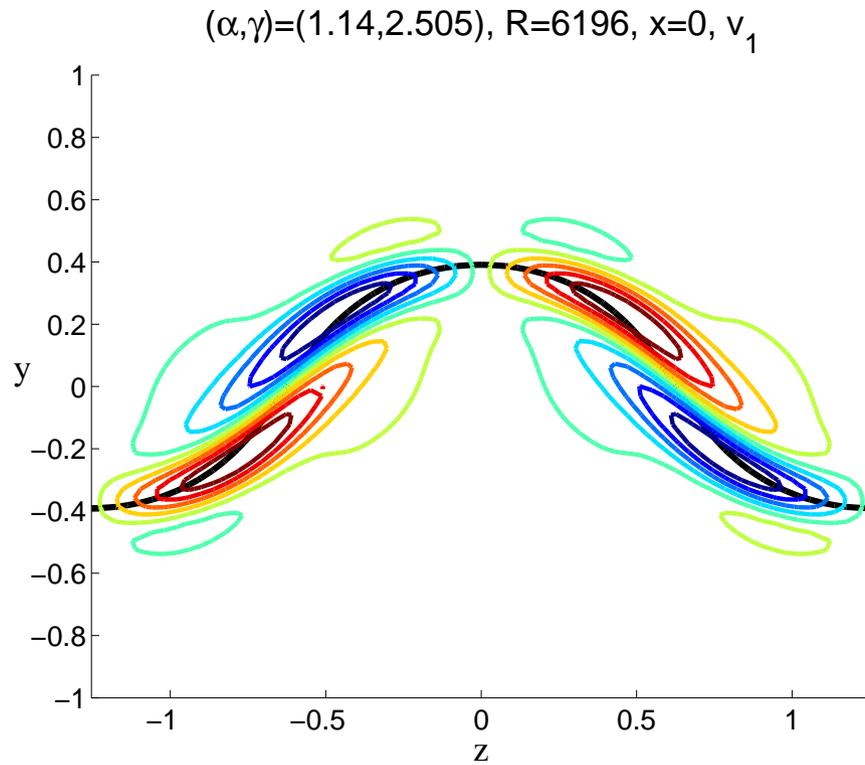
Structure of Lower branch @ $R = 6200$ (RRC)



u_0 large, $Q = \nabla^2 p / 2$ small

Waleffe & Wang, 2004, 2005

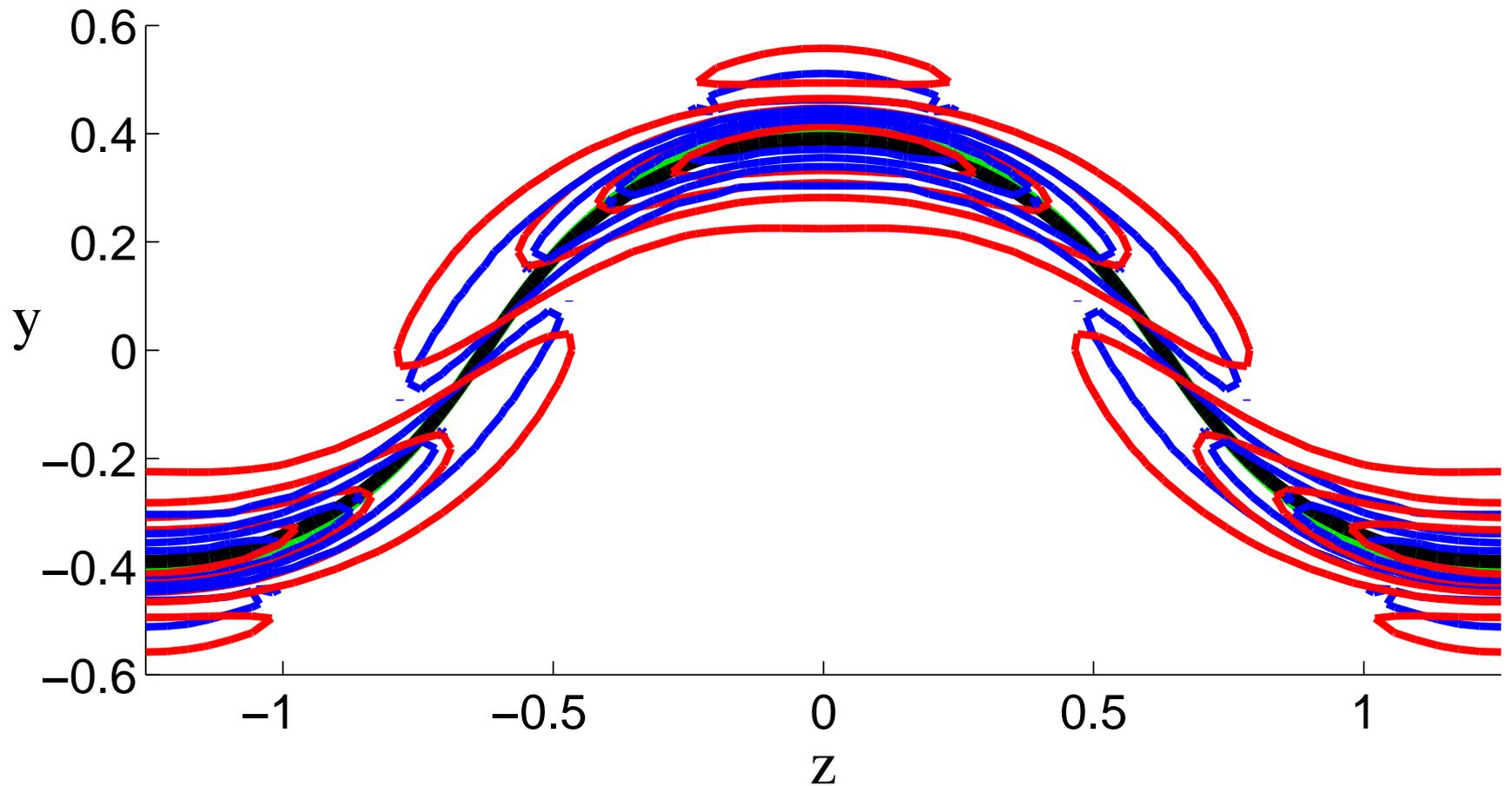
LBS 1st harmonic @ $R = 6200$ (RRC)



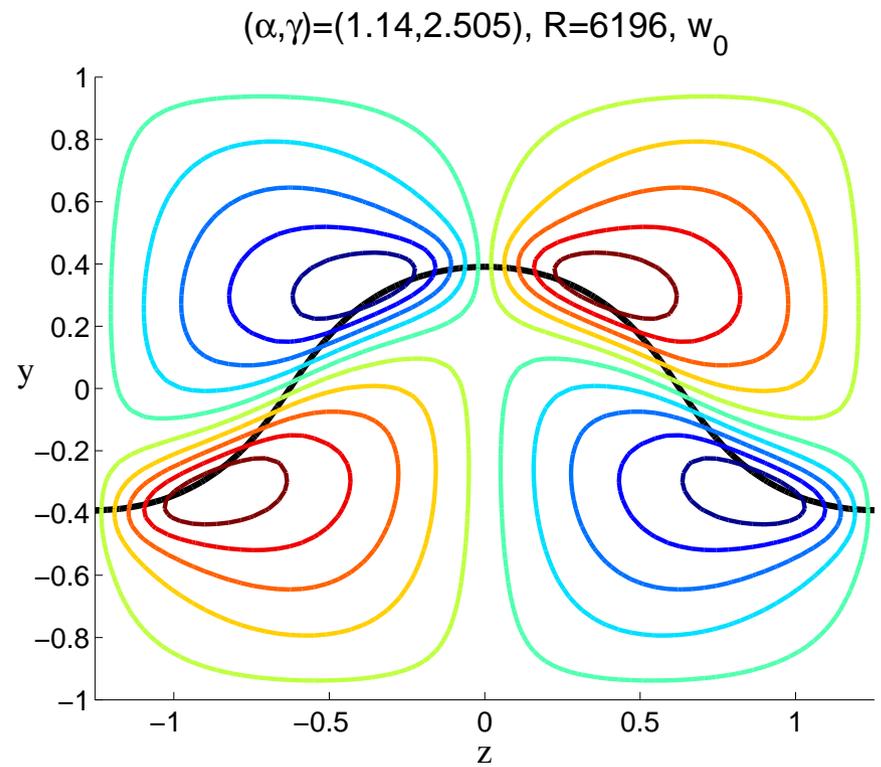
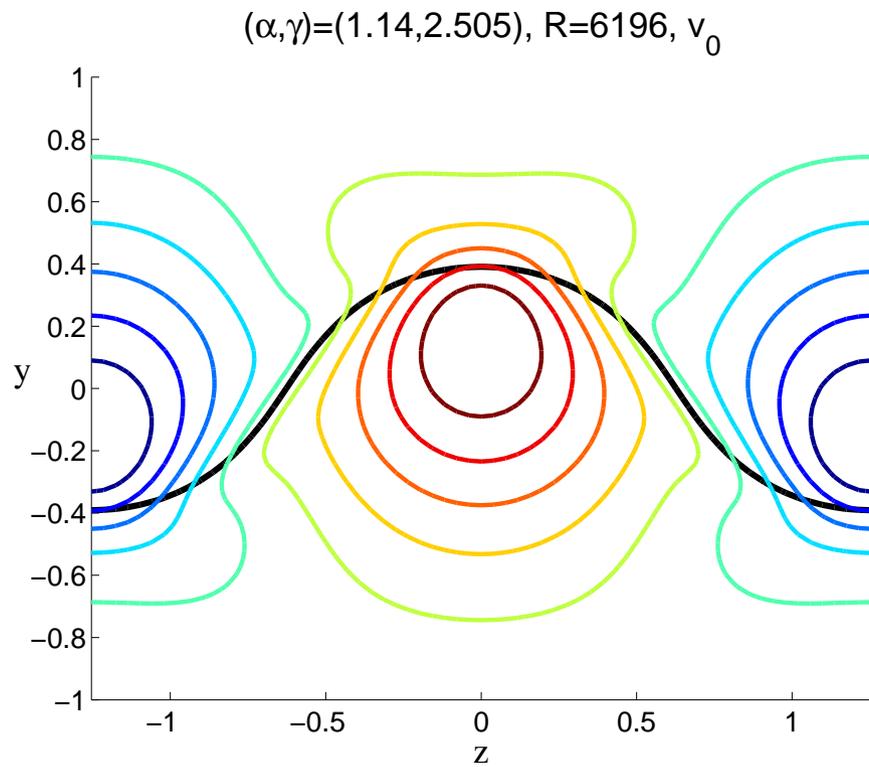
Critical layer! $u_0(y, z) - c = 0$

LBS 1st harmonic @ $R = 6196, 31599$

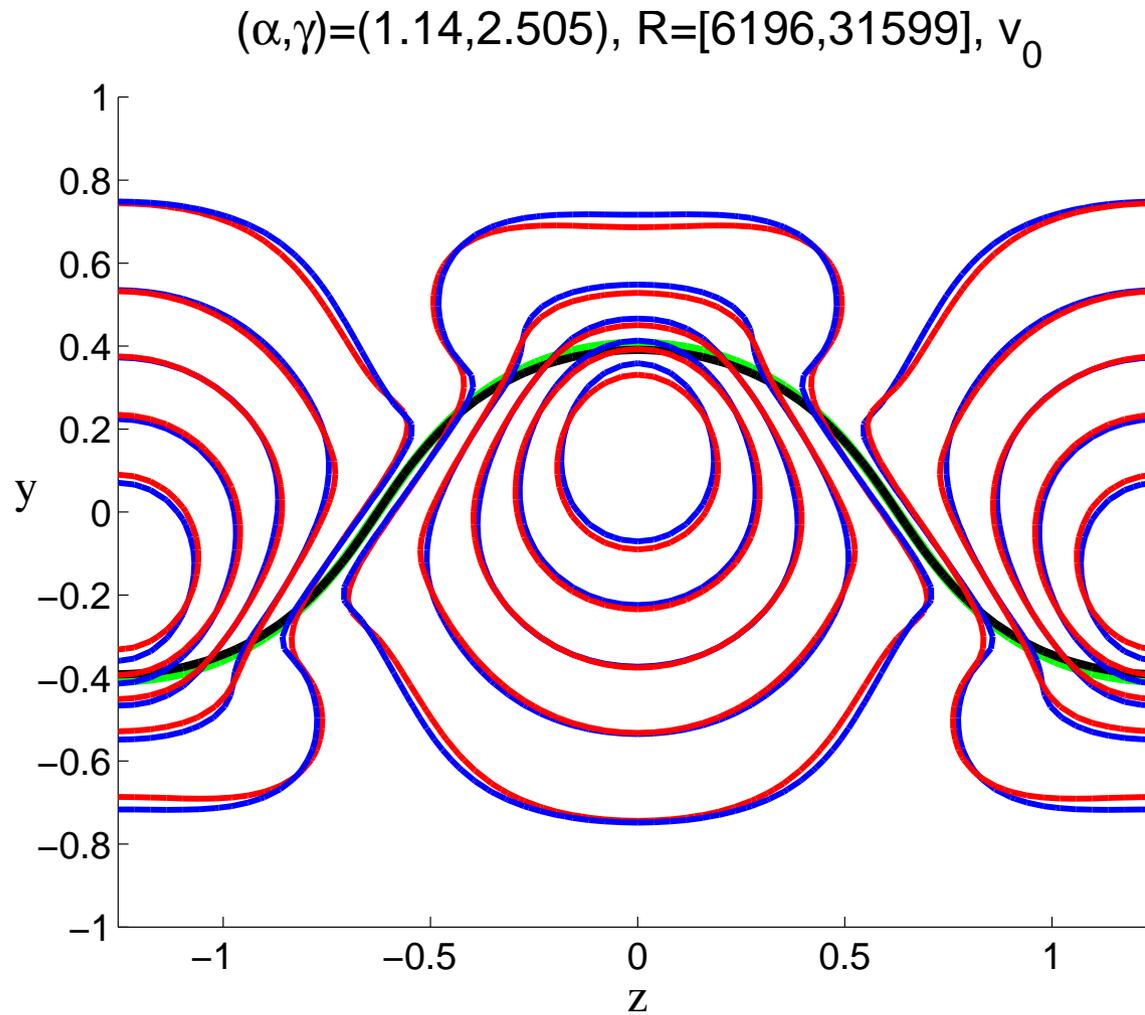
$(\alpha, \gamma) = (1.14, 2.505)$, $R = [6196, 31599]$, $x = 0$, w_1



LBS Rolls @ $R = 6200$ (RRC)



Structure of LBS @ $R = 6196, 31599$



SSP exact as $R \rightarrow \infty$ (but...)

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←

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- \approx Benney's 'Mean Flow-First Harmonic Theory'
(inviscid wavepackets, $\epsilon \longrightarrow$ viscous traveling wave, $1/R$)

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- \approx Benney's 'Mean Flow-First Harmonic Theory'
(inviscid wavepackets, $\epsilon \longrightarrow$ viscous traveling wave, $1/R$)
- but... 2D Critical Layer $u_0(y, z) - c = 0$
complicates scaling and asymptotics

SSP asymptotics: coupled 2D modes

$$(\partial_t + \mathbf{v} \cdot \nabla) \mathbf{v} + \nabla p = \frac{1}{R} \nabla^2 \mathbf{v}, \quad \nabla \cdot \mathbf{v} = 0$$

as $R \rightarrow \infty$, there are SF TWS

$$\begin{aligned} \mathbf{v}(x, y, z, t) &\sim u_0(y, z) \hat{\mathbf{x}} \\ &+ \left(v_0(y, z) \hat{\mathbf{y}} + w_0(y, z) \hat{\mathbf{z}} \right) \\ &+ e^{i\alpha(x-ct)} \hat{\mathbf{v}}_1(y, z) + c.c. \end{aligned}$$

SSP asymptotics: leading order eqns

- advection-diffusion of streaky flow $u_0(y, z)$:

$$v_0 \frac{\partial u_0}{\partial y} + w_0 \frac{\partial u_0}{\partial z} = \frac{1}{R} \nabla^2 u_0$$

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- Streamwise rolls: $v_0, w_0 \sim 1/R \Rightarrow u_0(y, z)$

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$$\frac{1}{R} \nabla^4 \Psi_0 = J(\nabla^2 \Psi_0, \Psi_0)$$

Rolls decoupled from streaky flow

$$J(A, B) = \partial_y A \partial_z B - \partial_z A \partial_y B$$

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$$v_0 \frac{\partial u_0}{\partial y} + w_0 \frac{\partial u_0}{\partial z} = \frac{1}{R} \nabla^2 u_0 - \frac{\partial \overline{u_1 v_1^x}}{\partial y} - \frac{\partial \overline{u_1 w_1^x}}{\partial z}$$

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$$\frac{1}{R} \nabla^4 \Psi_0 = J(\nabla^2 \Psi_0, \Psi_0) + \frac{\partial^2}{\partial y \partial z} (\overline{v_1^2}^x - \overline{w_1^2}^x) + \left(\frac{\partial^2}{\partial z^2} - \frac{\partial^2}{\partial y^2} \right) \overline{v_1 w_1}^x$$

1st harmonic feeds on streaks to sustain rolls

$$v_1 \sim 1/R \quad v_0, w_0 \sim 1/R$$

SSP asymptotics: leading order eqns

- advection-diffusion of streaky flow $u_0(y, z)$:

$$v_0 \frac{\partial u_0}{\partial y} + w_0 \frac{\partial u_0}{\partial z} = \frac{1}{R} \nabla^2 u_0 - \frac{\partial \overline{u_1 v_1^x}}{\partial y} - \frac{\partial \overline{u_1 w_1^x}}{\partial z}$$

- Streamwise rolls: $v_0(y, z) = \partial_z \Psi_0$, $w_0(y, z) = -\partial_y \Psi_0$

$$\frac{1}{R} \nabla^4 \Psi_0 = J(\nabla^2 \Psi_0, \Psi_0) + \frac{\partial^2}{\partial y \partial z} (\overline{v_1^2}^x - \overline{w_1^2}^x) + \left(\frac{\partial^2}{\partial z^2} - \frac{\partial^2}{\partial y^2} \right) \overline{v_1 w_1}^x$$

- First Harmonic neutrally stable $\mathbf{v}_1 = e^{i\alpha x} \hat{\mathbf{v}}_1(y, z) + c.c.$

$$(\mathbf{v}_0 - c\hat{\mathbf{x}}) \cdot \nabla \mathbf{v}_1 + \mathbf{v}_1 \cdot \nabla \mathbf{v}_0 + \nabla p_1 = \frac{1}{R} \nabla^2 \mathbf{v}_1, \quad \nabla \cdot \mathbf{v}_1 = 0$$

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- The neglected lower branch states:
'large' scale, 2D self-sustained critical layer
drag only 10-20% higher than laminar!
control?